

# Aircraft Structural Mass Property Prediction Using Conceptual-Level Structural Analysis

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# Summary

- ELAPS is a conceptual-level aircraft structural analysis tool uniquely situated between empirical methods and finite element analysis
- The ELAPS-based mass property analysis process enables a low cost, high fidelity multidisciplinary approach to conceptual design
- Stochastic analysis techniques provide a robust and efficient framework for managing uncertainty in conceptual design



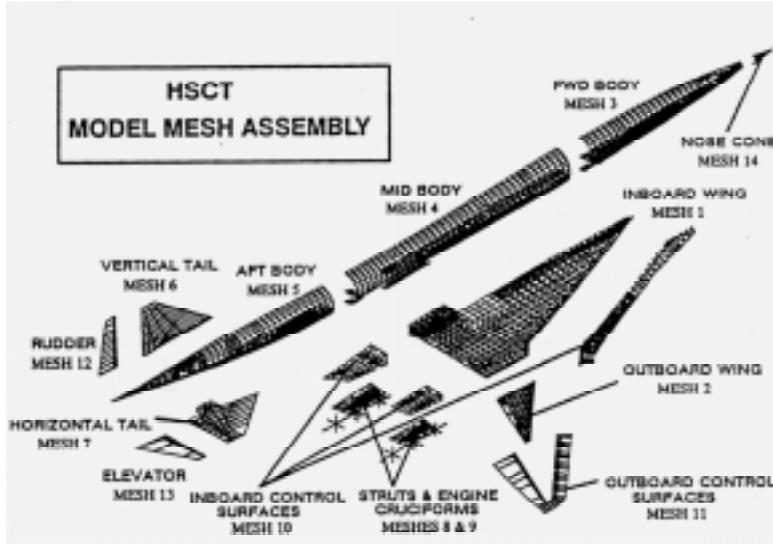
# Overview

- Structural mass property prediction
- ELAPS
- ELAPS structural models
- Non-optimal mass
- Stochastic analysis and design
- ELAPS stochastic mass property analysis
- Case study: ERAST Proof-of-Concept
- Summary & Conclusions



# Structural Mass Property Prediction

$$Weight_{fuselage} = 1.35 \cdot \left[ X_L \cdot \left( \frac{W_{fuselage} + D_{fuselage}}{2} \right) \right]^{1.28} \cdot \\ (1.0 + 0.05 \cdot N_{engine\_fuselage}) \cdot (1.0 + 0.38 \cdot W_{cargo\_fuselage}) \cdot N_{fuselage}$$



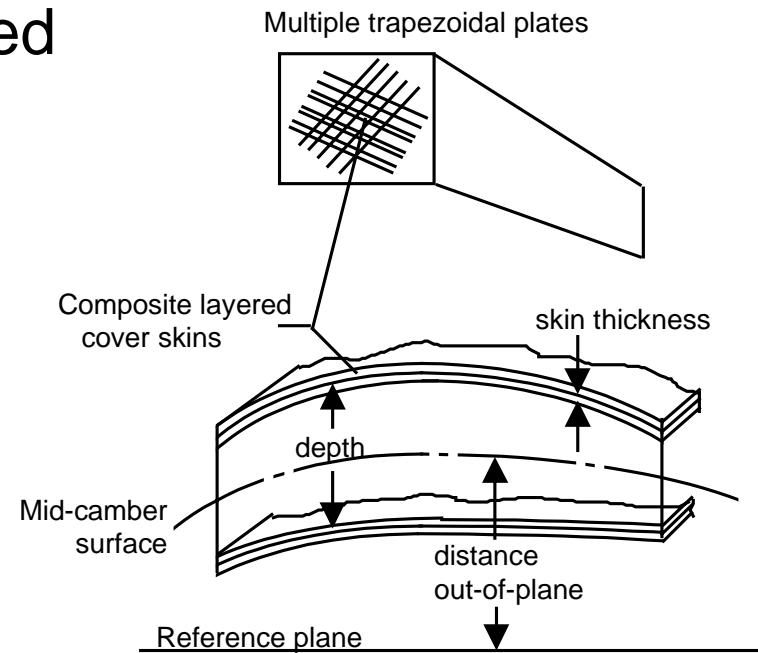
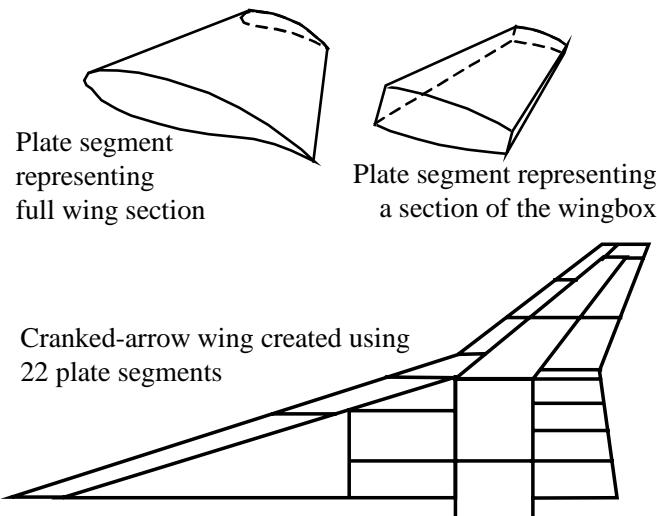
(from Mitchell, 1993)

- Empirical equations
- Empirically-augmented physical methods
- Finite Element Analysis (FEA)
- Algorithmic Mass Factoring Method (AM-FM, Boeing)
- Mock-ups & prototypes



# ELAPS

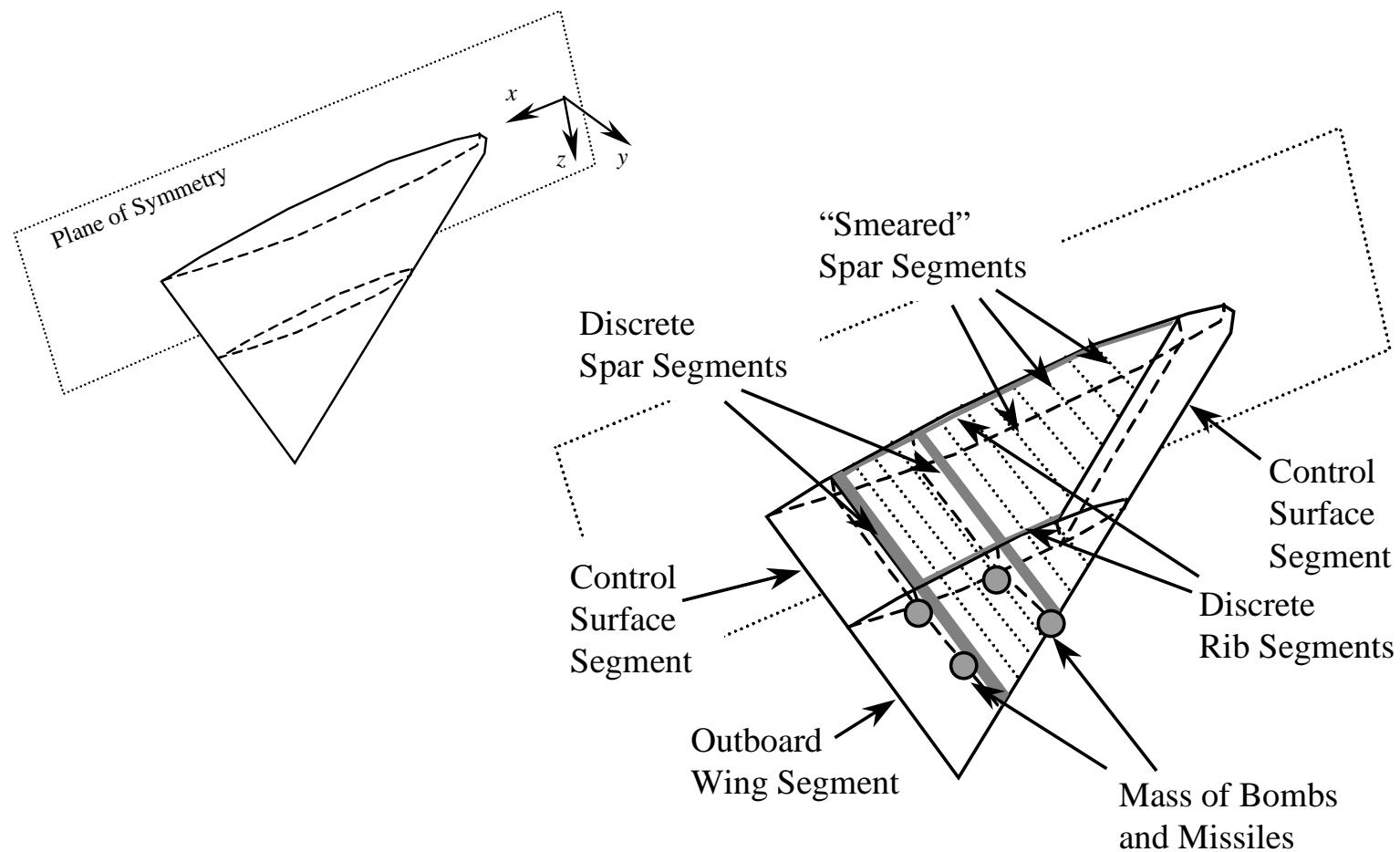
- ELAPS = Equivalent LAminated Plate Solution
- FEA → elements
- ELAPS → segments



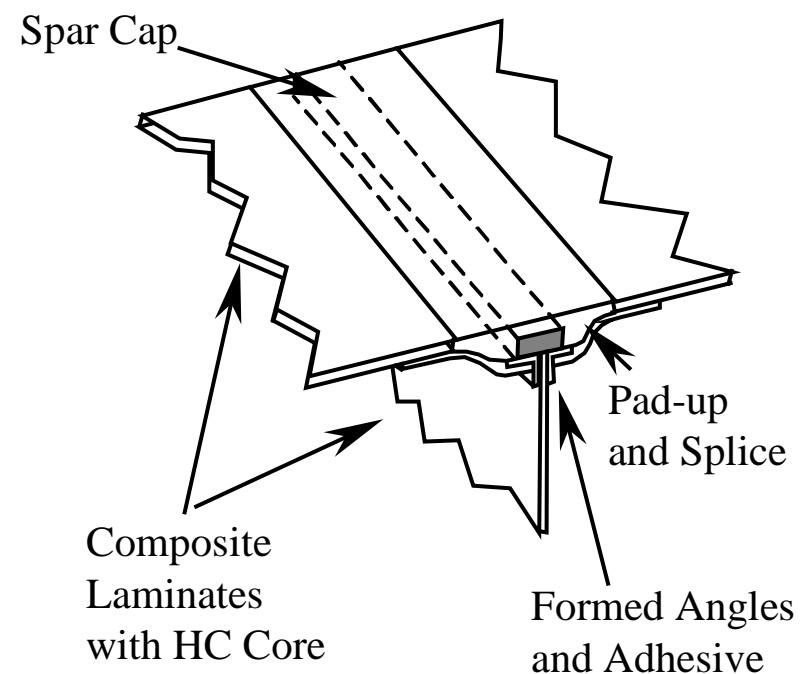
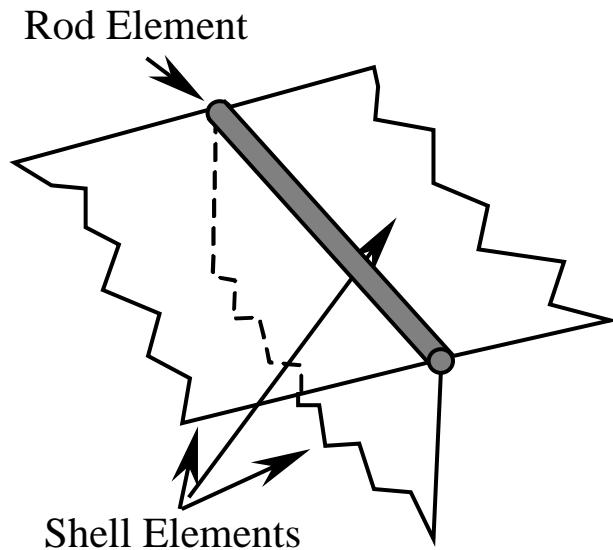
- Fast modeling
- Low computational cost



# ELAPS Structural Models



# Non-Optimal Mass

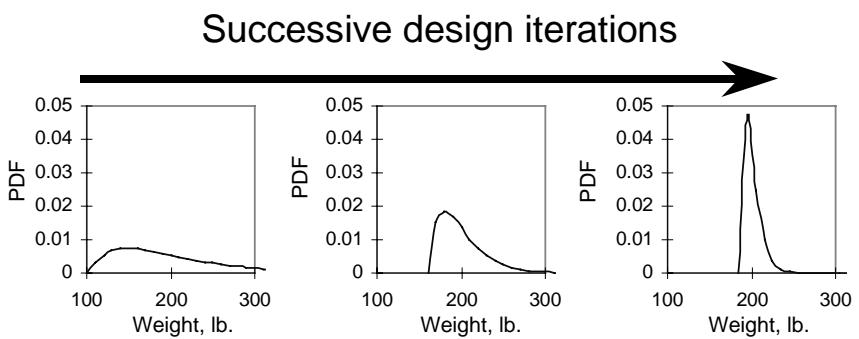
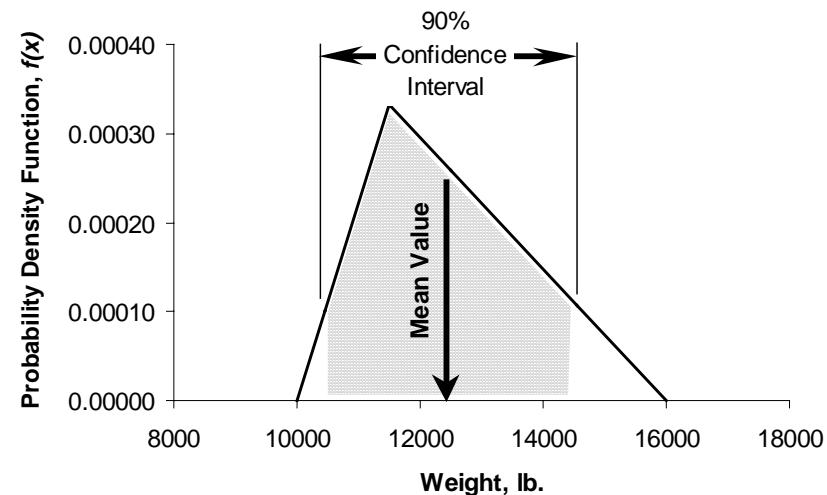


$$NOMF = \frac{as-built\ mass}{ideal\ mass}$$



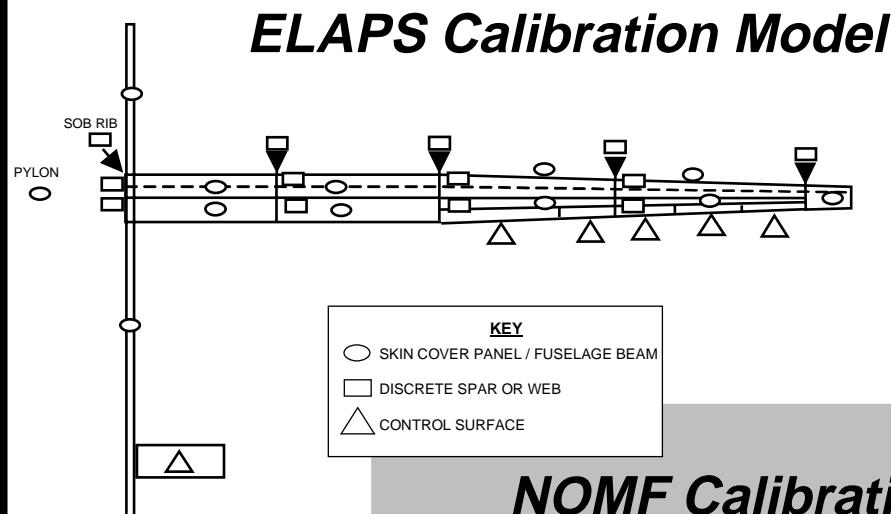
# Stochastic Analysis and Design

- Management of uncertainty
- Robustness
- Risk Assessment
- Learning

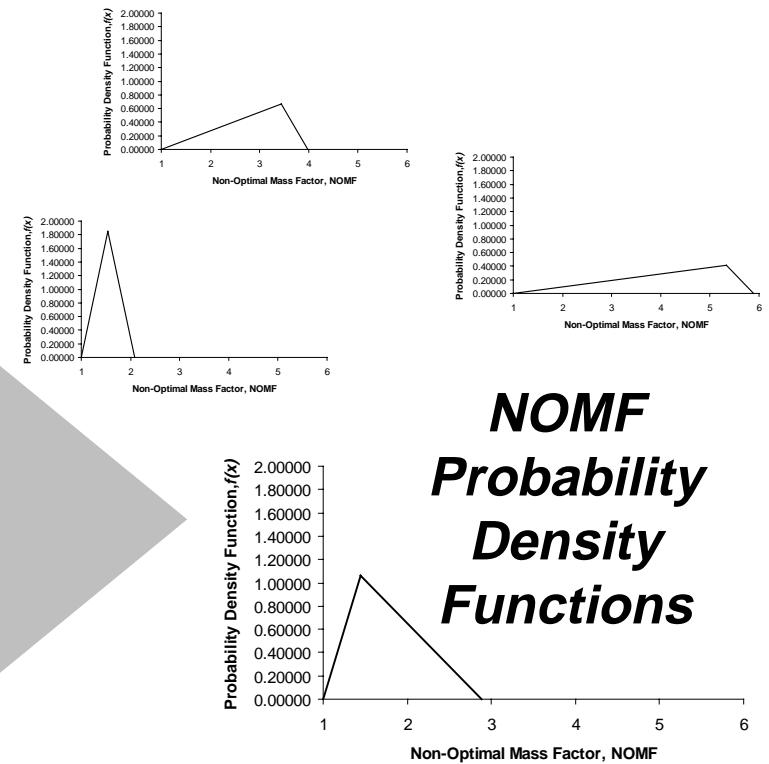


# ELAPS Stochastic Mass Property Analysis

## *NOMF Uncertainty*

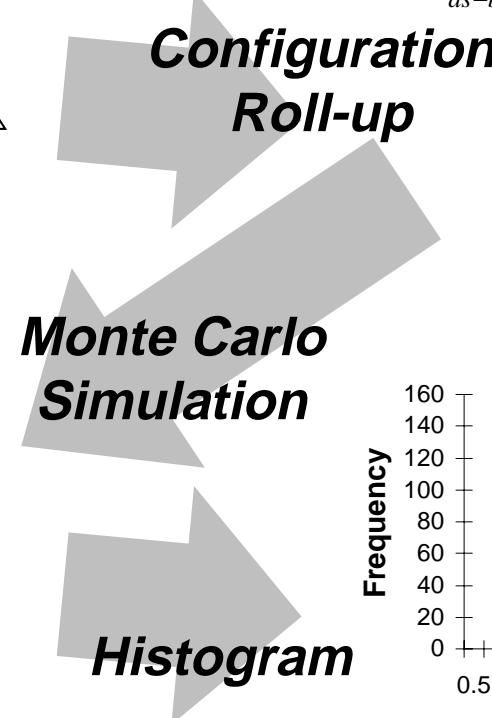
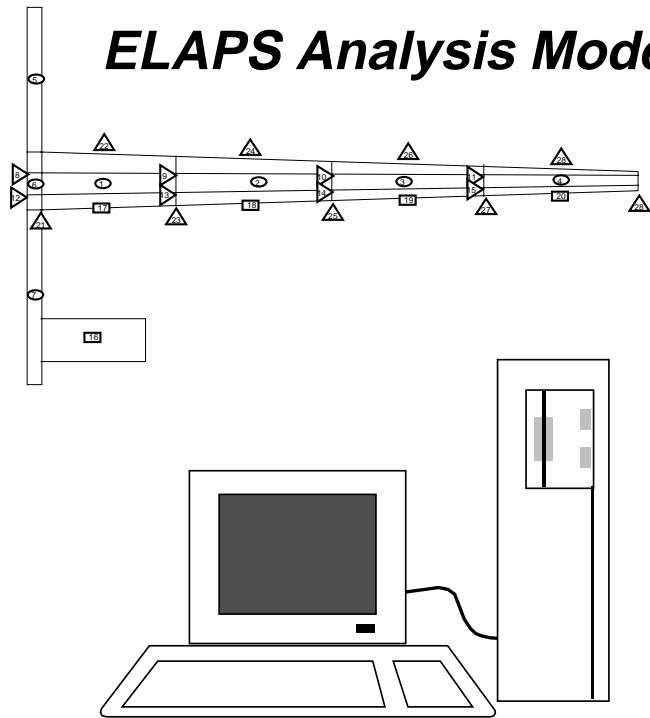


**NOMF Calibration**

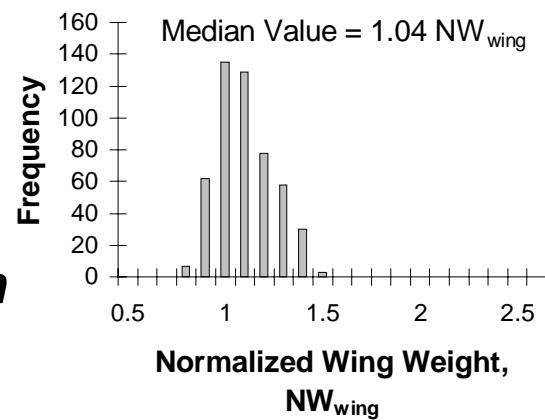


# ELAPS Stochastic Mass Property Analysis

## *Monte Carlo Simulation*

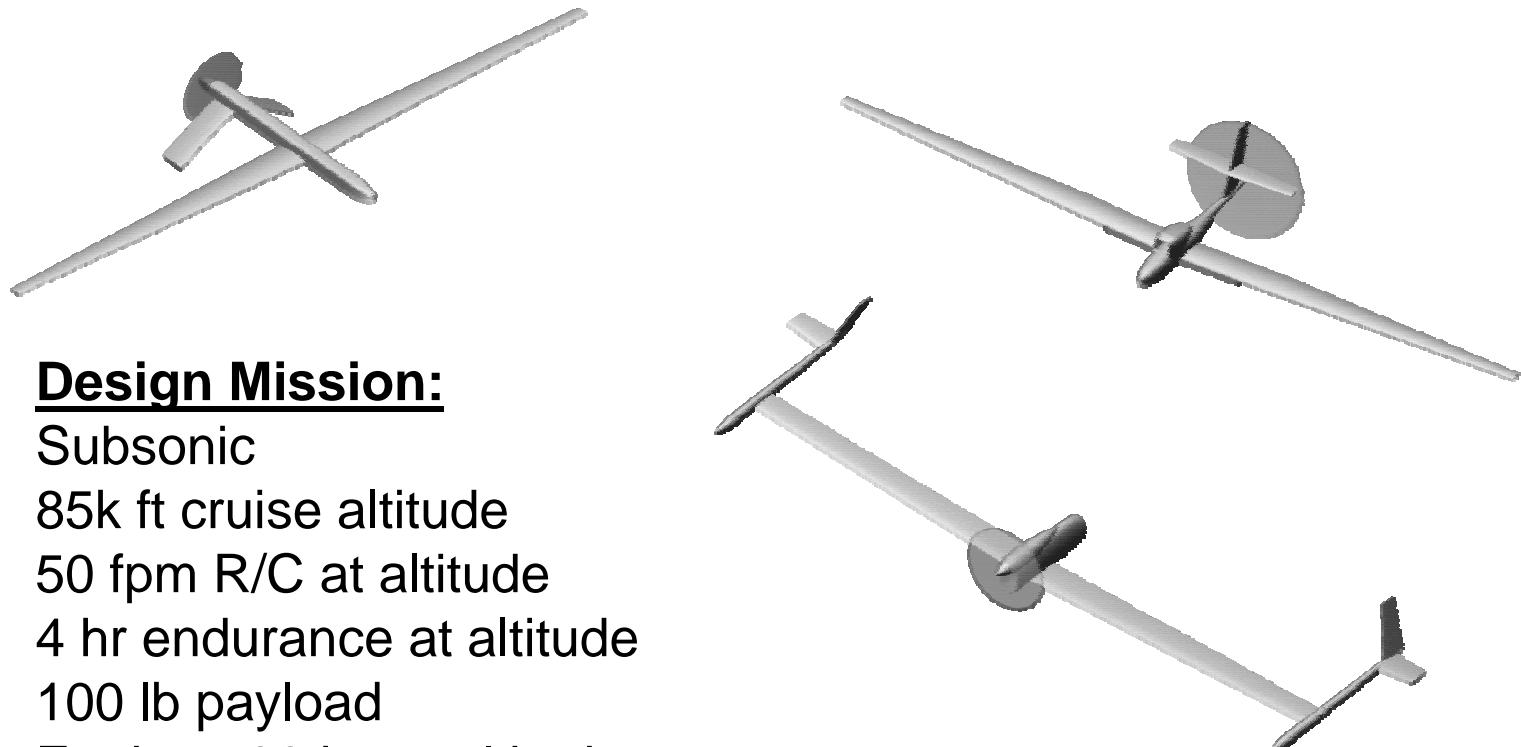


$$W_{as-built} = \sum_{i=1}^{Wing\_Skin\_Components} W_{Ideal_i} \cdot NOMF_i + \sum_{i=1}^{Wing\_Spar\_Components} W_{Ideal_i} \cdot NOMF_i + \sum_{i=1}^{Wing\_Rib\_Components} W_{Ideal_i} \cdot NOMF_i + \dots$$



# Case Study: ERAST Proof-of-Concept

## *Design Proposals*



### Design Mission:

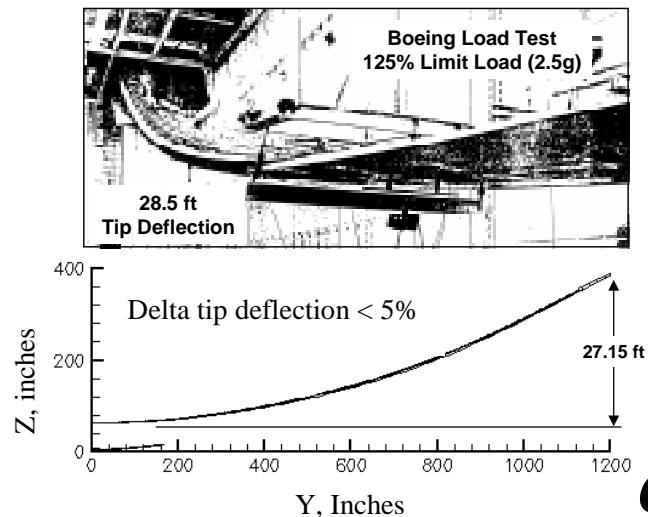
Subsonic  
85k ft cruise altitude  
50 fpm R/C at altitude  
4 hr endurance at altitude  
100 lb payload  
Engine: 80 hp at altitude



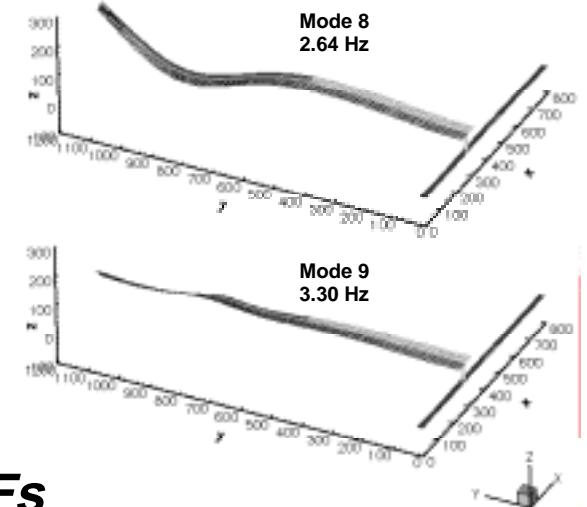
# Case Study: ERAST Proof-of-Concept *ELAPS Calibration Model*

*Boeing Condor UAV*

## *Static Deflection Comparison*



## *Flutter Mode Comparisons*



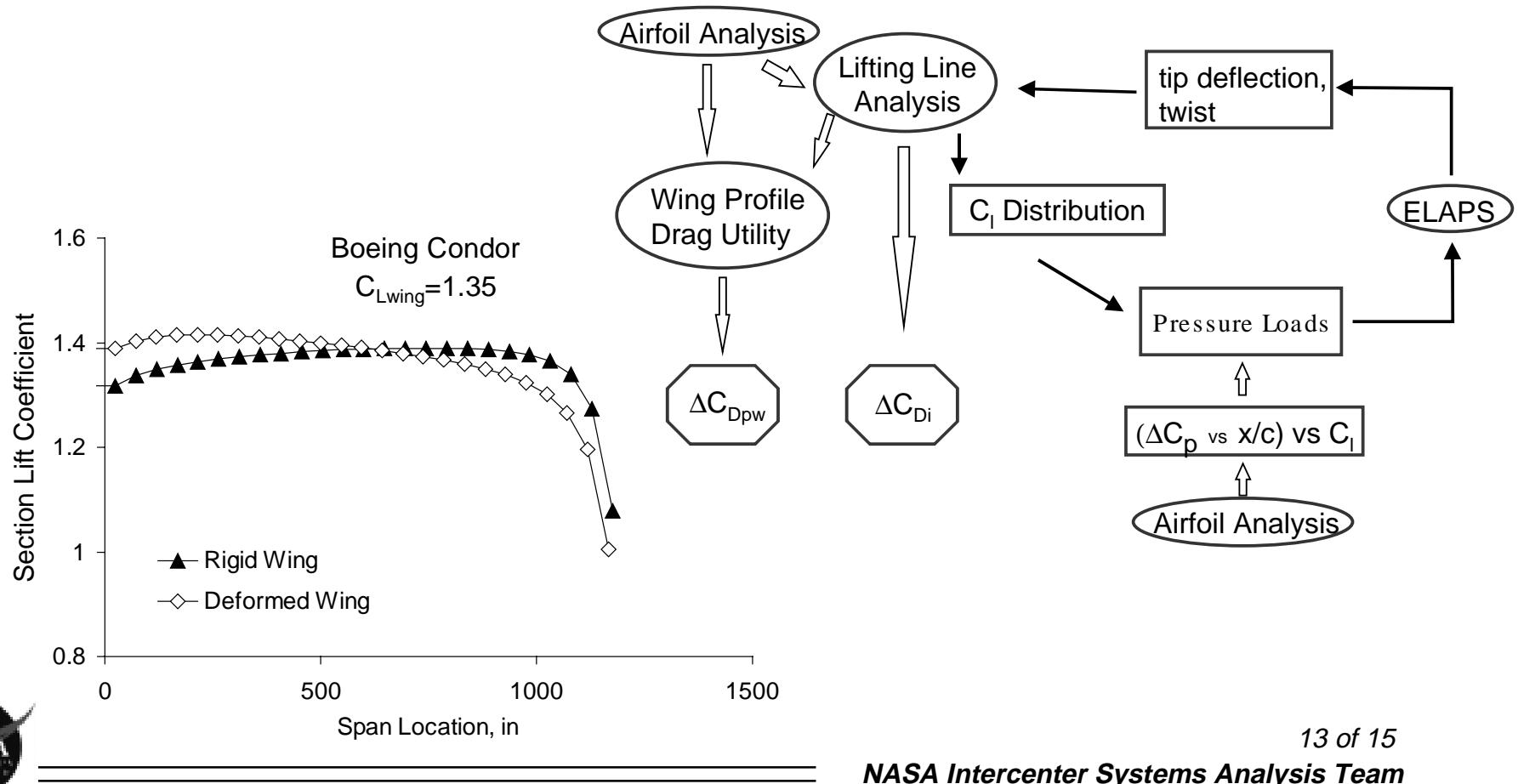
*Group Weight Statement*

*Component NOMF PDFs*

$$NOMF_i = f(x)$$



# Case Study: ERAST Proof-of-Concept *Aeroelasticity Analysis*



# Case Study: ERAST Proof-of-Concept

## *Mass Property Analysis*

***ELAPS Model***

***Structural Sizing***

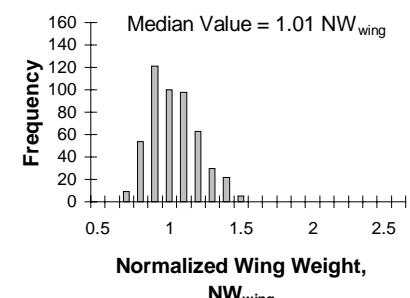
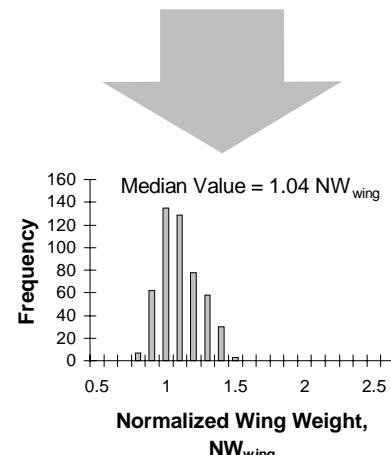
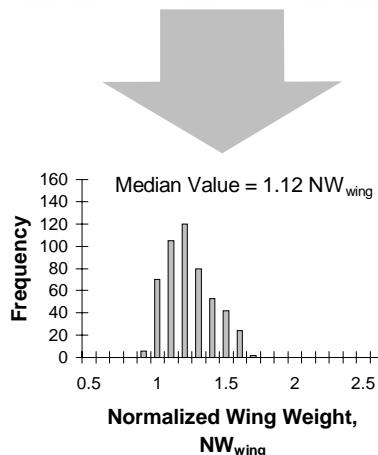
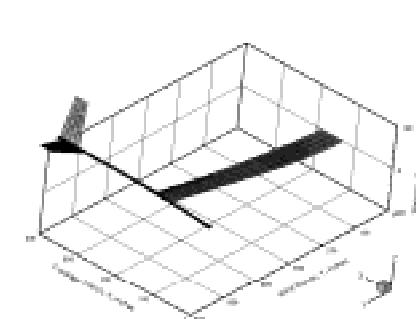
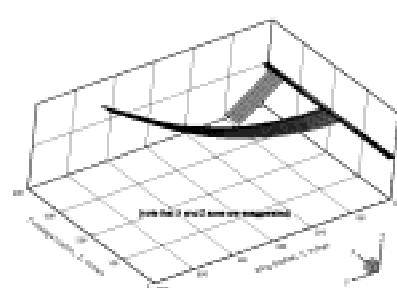
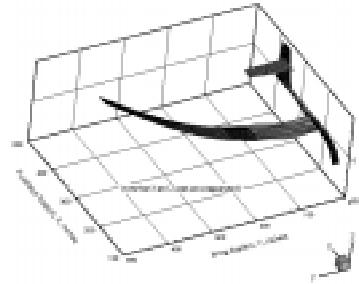
***Ideal Mass***

**$NOMF_i = f(x)$**

***As-built Mass***

***Monte Carlo***

***Histogram  
or  
PDF***



# Summary & Conclusions

- ELAPS is a structural analysis tool uniquely suited to multidisciplinary conceptual design
  - Enables high-fidelity structural behavior knowledge early in the design process (including aeroelasticity)
  - Speed of modeling and analysis significantly reduces design and analysis cycle time
- The ELAPS-based stochastic mass property analysis process facilitates weight risk assessment, especially in cases of advanced technology or unusual vehicle configuration

